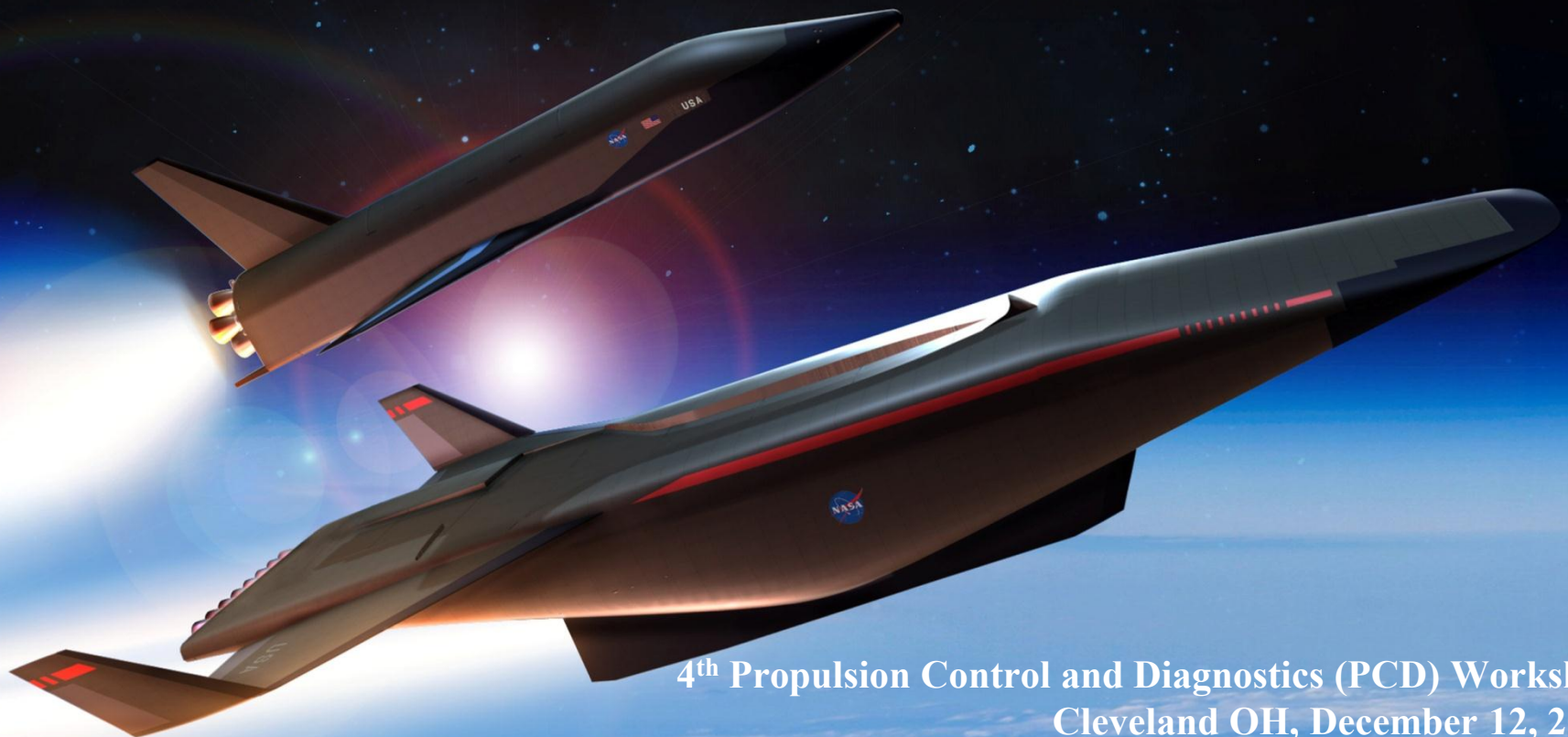




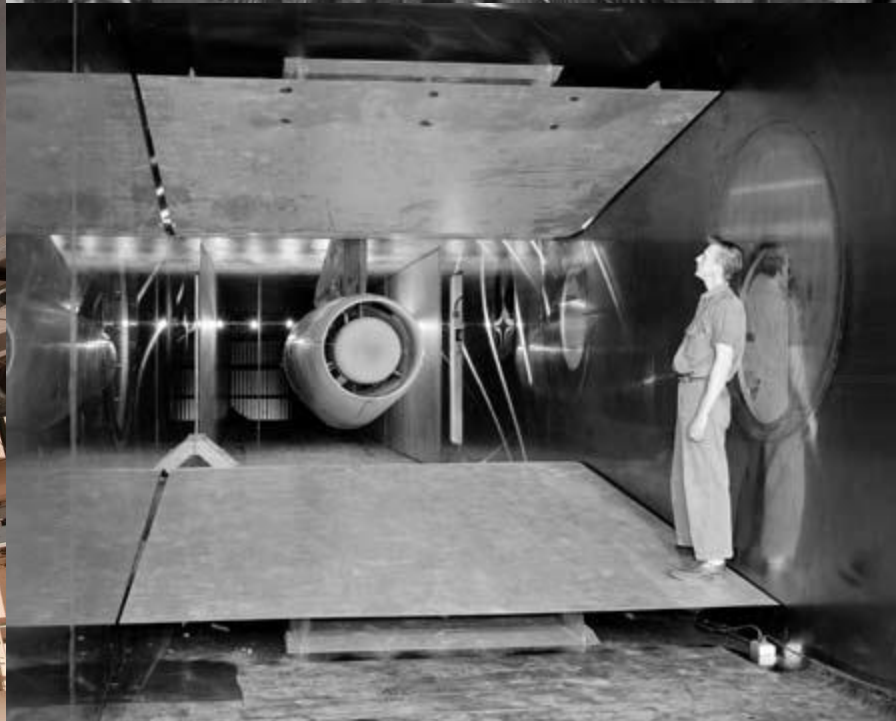
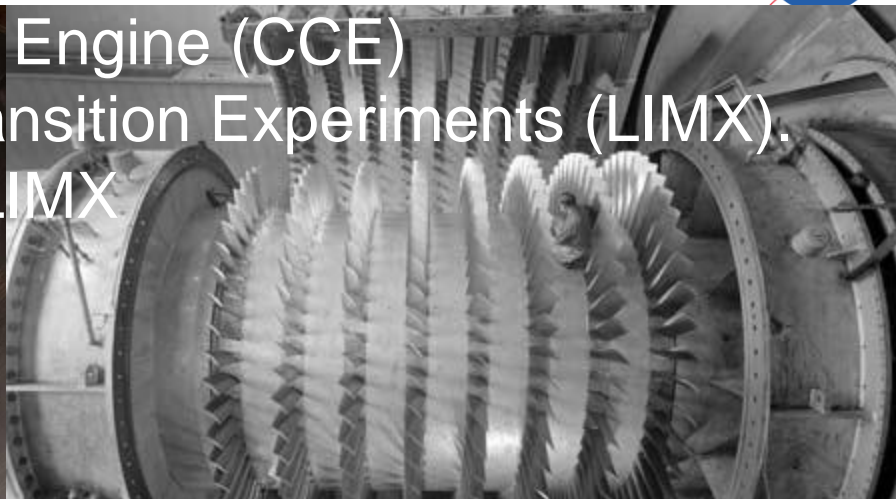
**Combined Cycle Engine (CCE) Large Scale Inlet for Mode Transition
Experiments (LIMX) Mode Transition Modeling and Control
Fundamental Aeronautics – Hypersonic Project**

**Thomas J. Stueber
NASA Glenn Research Center
Cleveland, Ohio**



**4th Propulsion Control and Diagnostics (PCD) Workshop
Cleveland OH, December 12, 2013**

Combined Cycle Engine (CCE) Large-Scale Inlet for Mode Transition Experiments (LIMX). CCE-LIMX



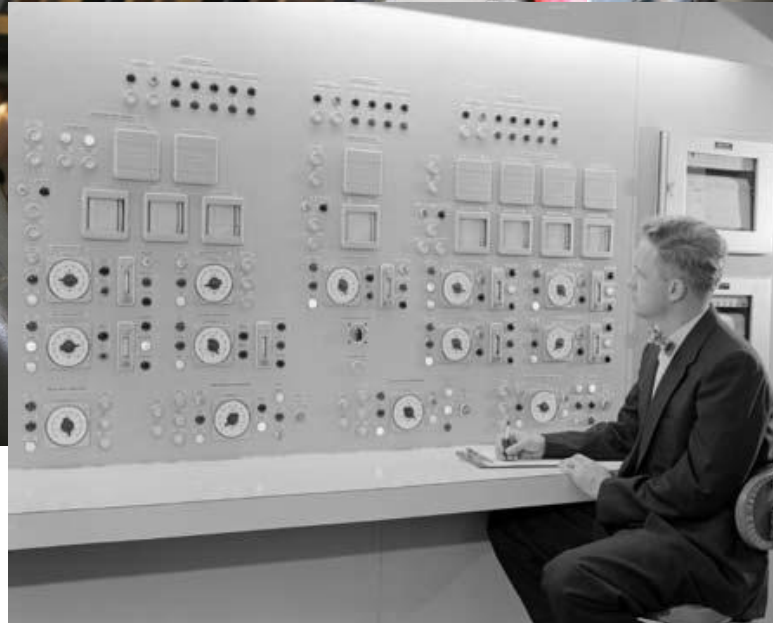


Team

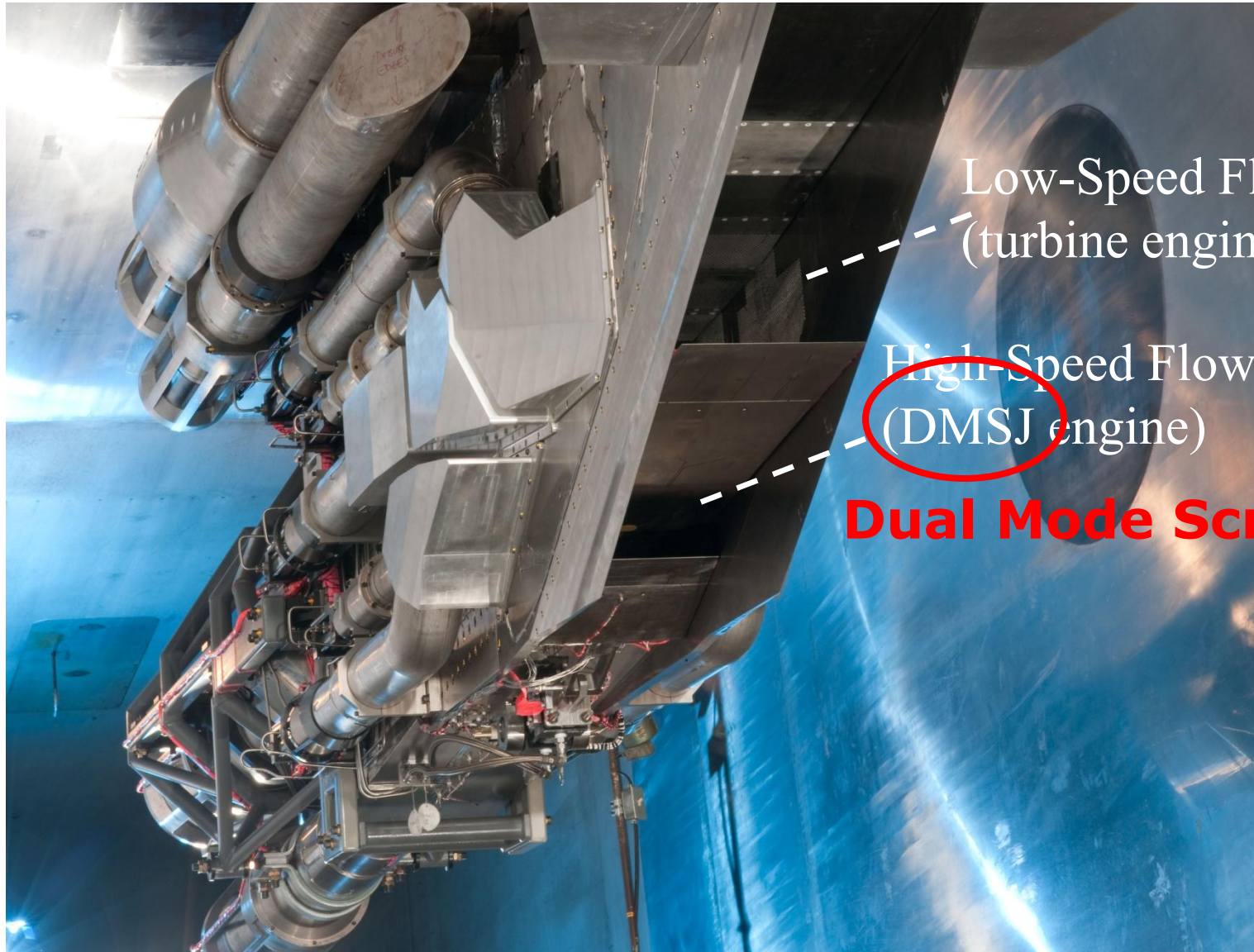
- **NASA GRC Research Directorate:**
 - Communication, Instrumentation, and Controls Division
 - Controls and Dynamics Branch (RHC)
 - Jeffrey T. Csank, Thomas J. Stueber, Randy Thomas
 - Digital Communications and Navigation (RHD)
 - Joseph A. Downey, Jennifer M. Nappier, Binh V. Nguyen
 - Aeropropulsion Division
 - Inlet & Nozzle Branch (RTE)
 - David Davis, Lancert E. Foster, Dave Saunders, John W. Slater
- **NASA GRC Engineering Directorate:**
 - Systems Engineering and Analysis Division
 - Propulsion & Control Systems Engineering (DSS)
 - Dzu K. Le, Daniel R. Vrnak
- **NASA GRC Facilities & Test Directorate:**
 - Testing Division, Wind Tunnel & Propulsion Test Branch.
- **Industry Partners**
 - TechLand Research Inc. Bobby W. Sanders, Lois J. Weir
 - Spiritech Advanced Products Inc., Eric J. Gamble, Dan Haid, Sal D'Alessandro, Rich DeFrancesco.

Outline

- Overview of CCE-LIMX Inlet System
- Control Affecters
- Control Design
- Phase-3 Test Plans
- Summary



CCE-LIMX Model

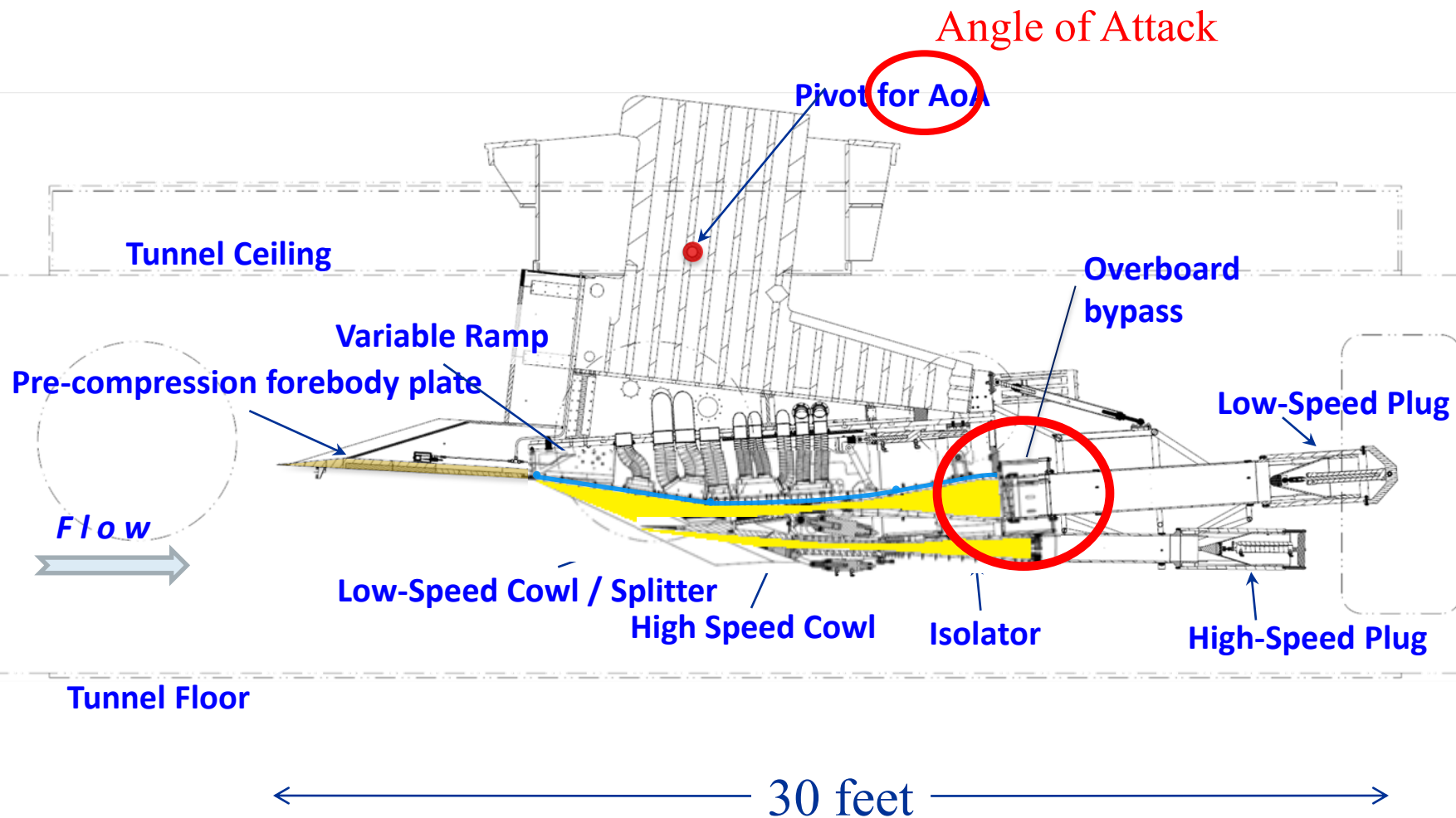


Low-Speed Flow Path
(turbine engine)

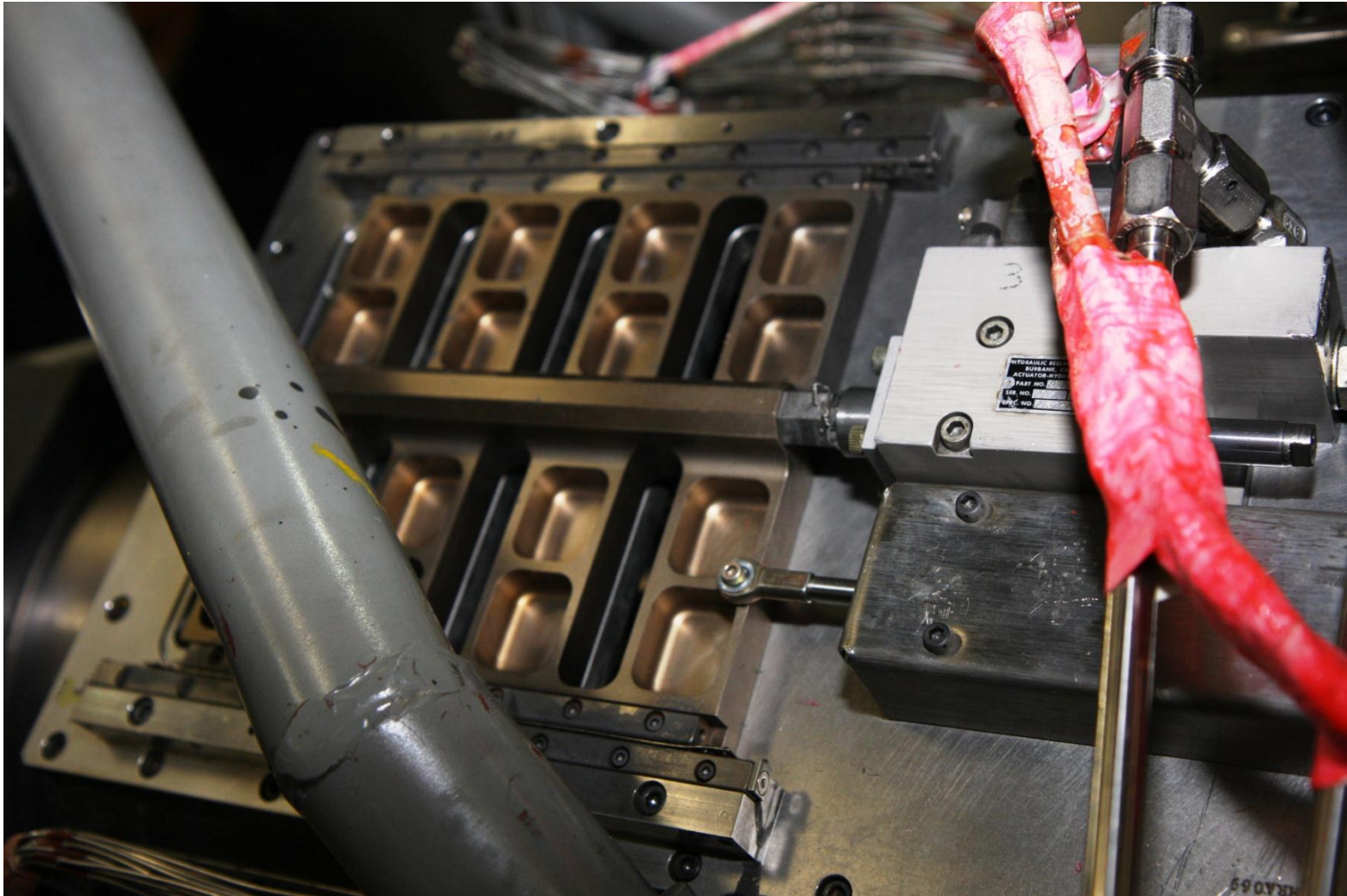
High-Speed Flow Path
(DMSJ engine)

Dual Mode Scramjet

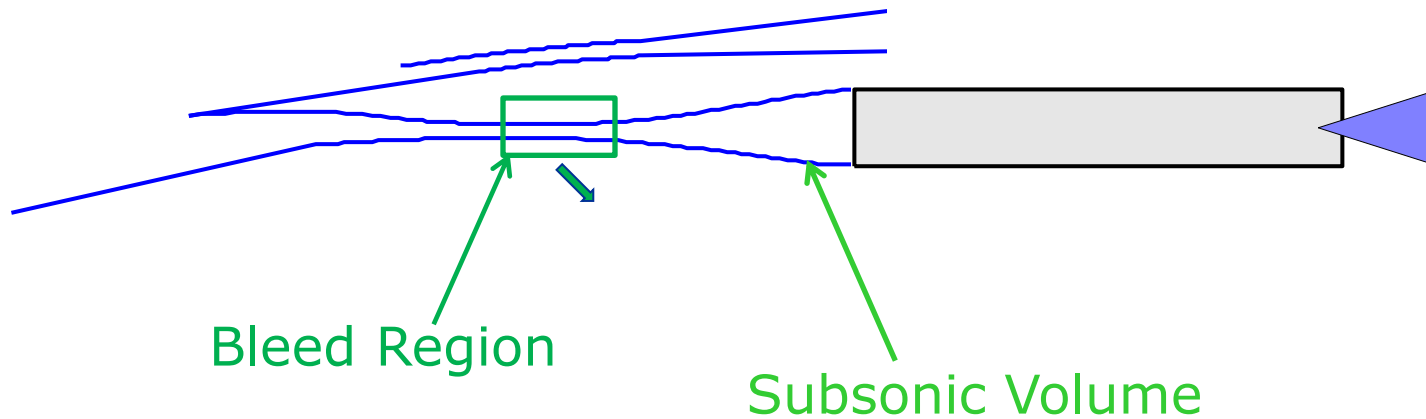
CCE-LIMX Model Features



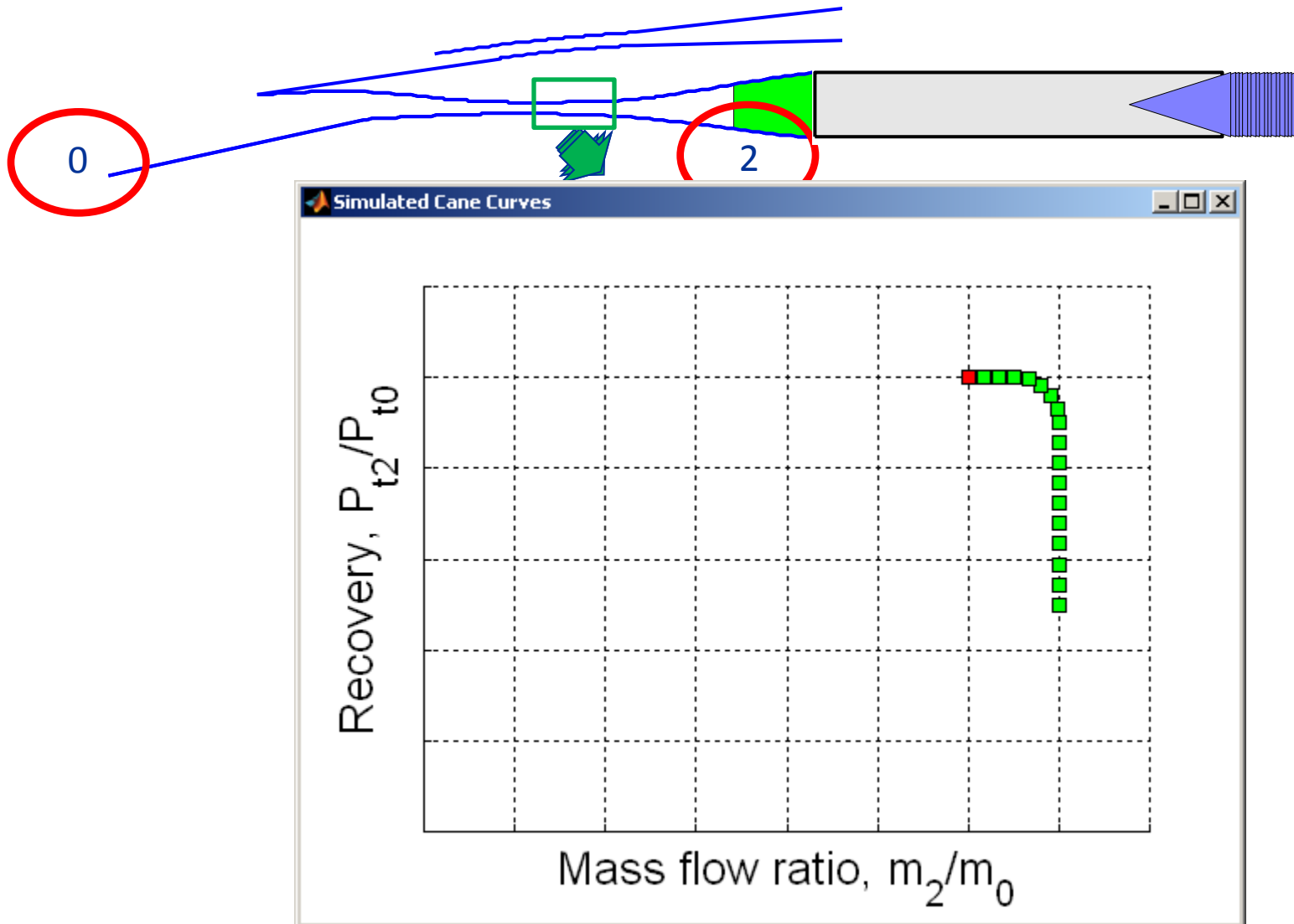
Control Affecters



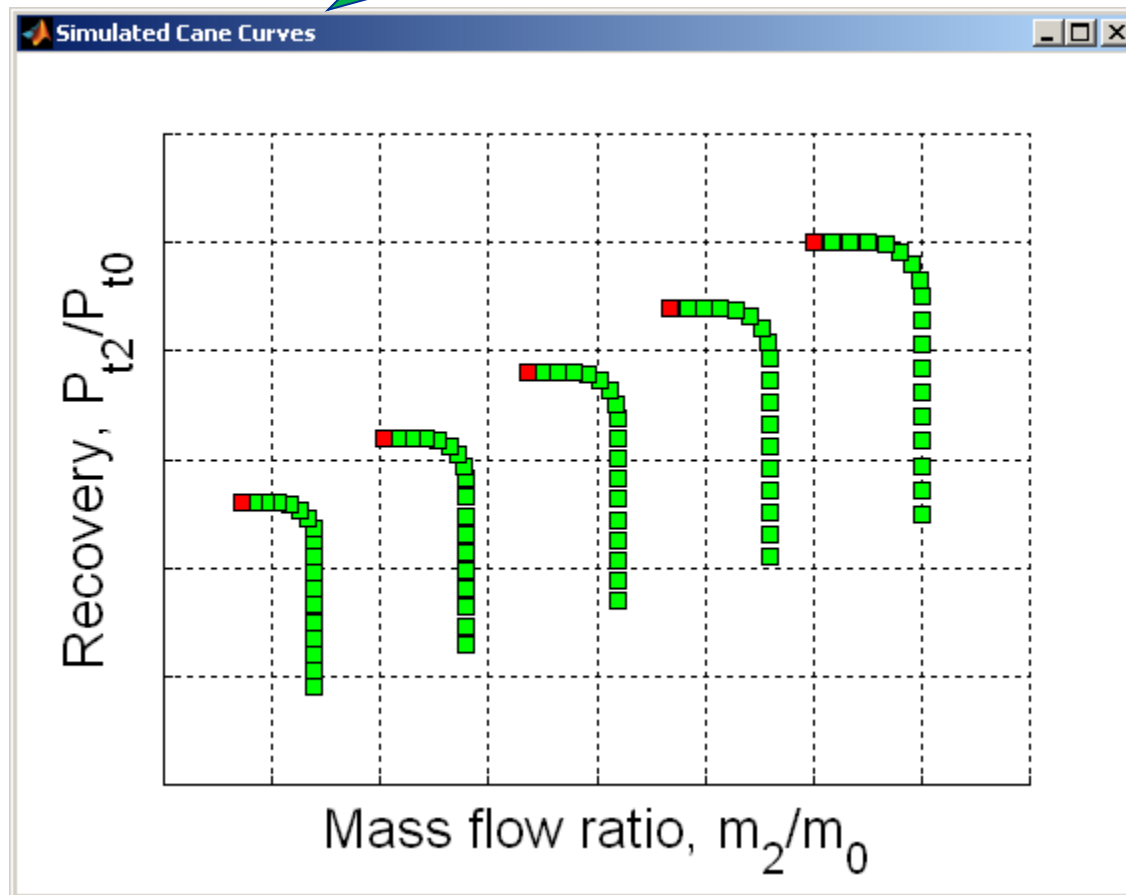
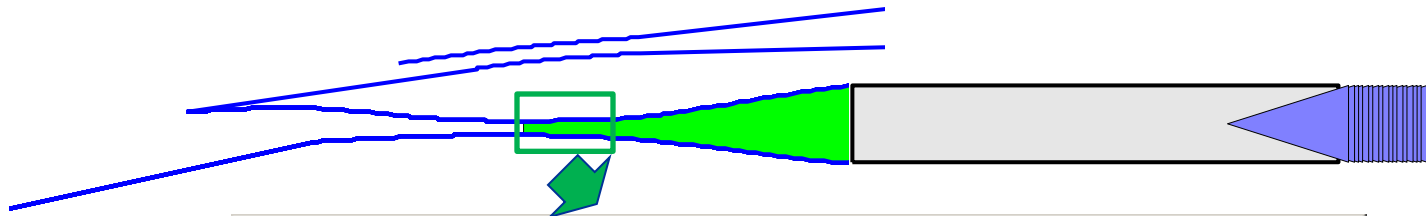
Phase 1: Inlet Characterization and Performance Testing



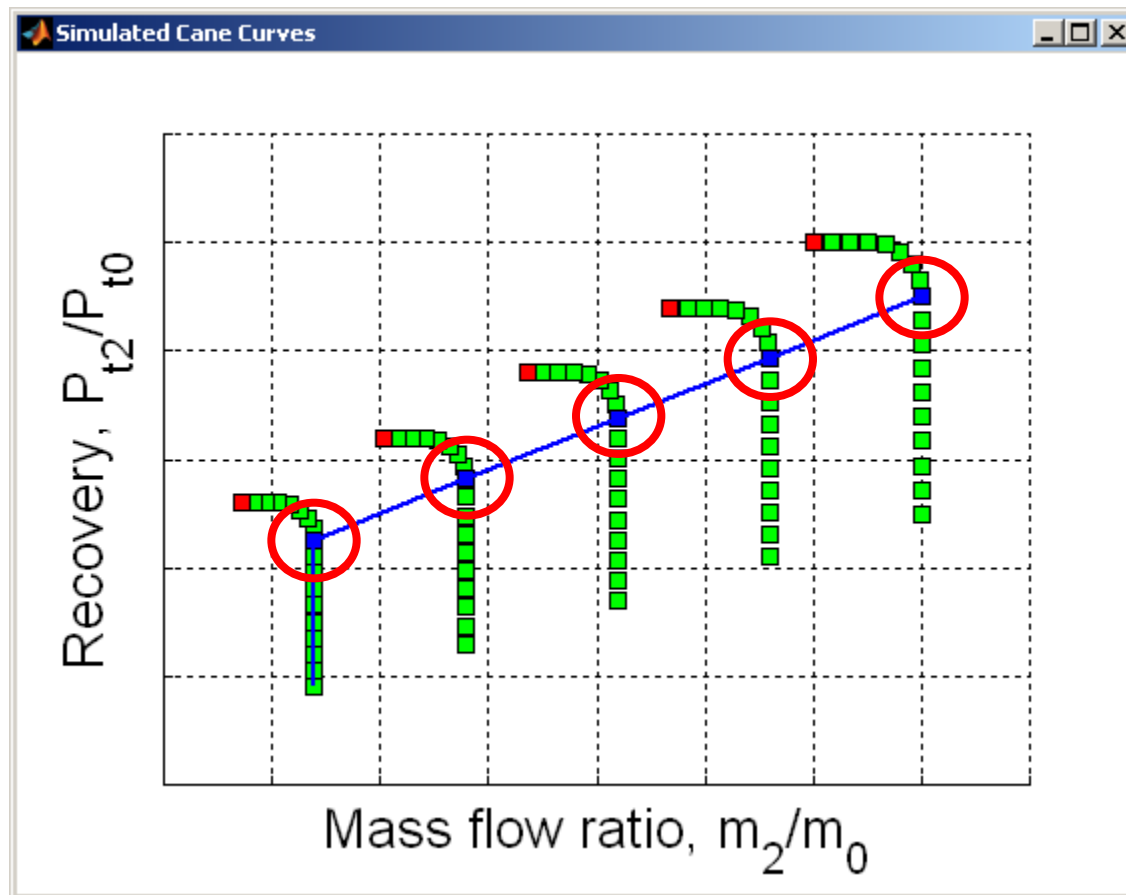
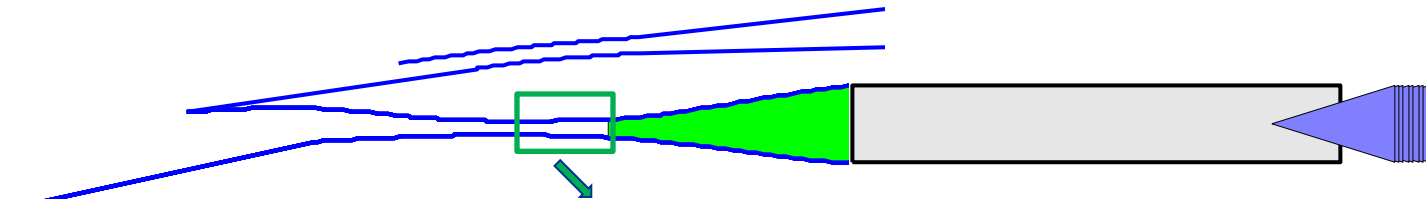
Phase 1: Inlet Characterization and Performance Testing



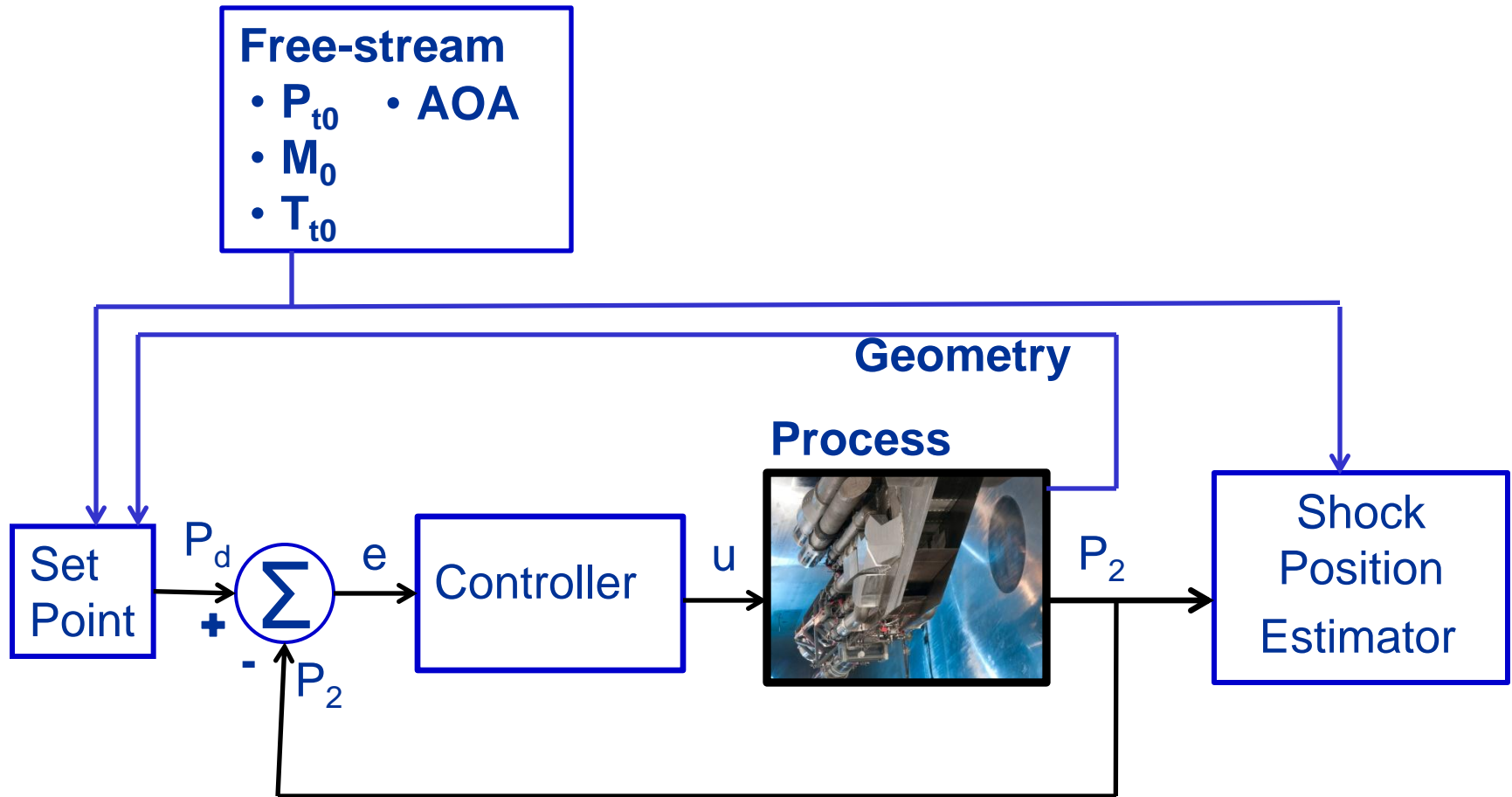
Phase 1: Inlet Characterization and Performance Testing



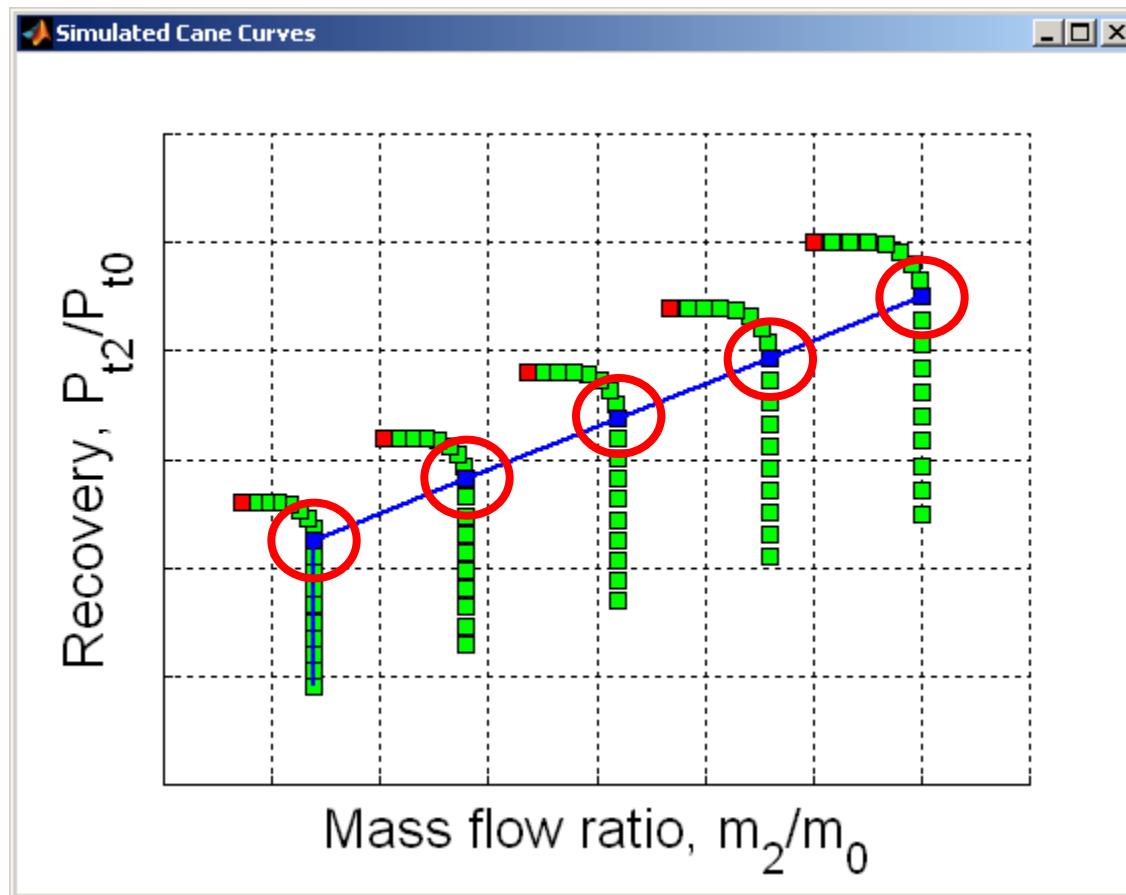
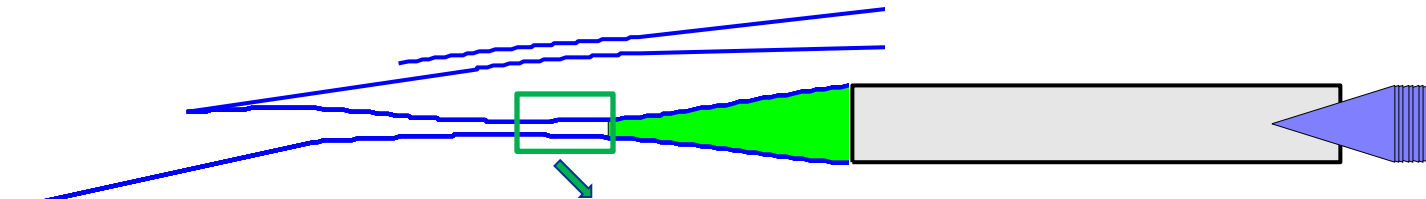
Phase 1: Inlet Characterization and Performance Testing



Control Design



Phase 1: Inlet Characterization and Performance Testing





Phase 2 Accomplishments

- Test matrix status Phase 2 Mach 4
 - 642 experiments identified, ~89 hrs
 - Main (LST1 and HST1) schedule—506 experiments, ~49 hrs
 - First alternate (LST1 and HST2) schedule—68 experiments, ~20 hrs
 - Second alternate (LST2 and HST2) schedule—68 experiments, ~20 hrs
 - Reduced matrix—393 experiments selected, ~29 hrs
 - Main schedule—378 experiments completed, 38.25 hrs
 - Alternates—0 experiments completed
 - Experiments:

• Step,	Sinusoidal-Sweep,	Sustained-Sinusoid
• Staircase,	Transient Stability Index (S_{it}),	
• Unstart,	Buzz,	Restart
- Test window: 8/29/2011 – 10/19/2011
- 11 run nights (data collection)



Phase-3 Plans

- Open-Loop mode transition along schedule
 - Transient stability index at operating points on schedule
- Closed-Loop mode transition along schedules
 - Perturbation testing at each operating point:
 - Step change
 - Sinusoidal sweep
 - Sinusoid pulse
 - Perturbation testing during closed-loop mode transition.



Summary

- Combined Cycle Engine (CCE) Large Scale Inlet for Mode Transition Experiments (LIMX)
 - Designed,
 - Built,
 - Installed in NASA GRC 10- x 10-foot SWT
 - Completed Phase-1 and Phase-2 testing
 - Preparing for Phase-3 testing.



Questions?